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# Coexistence between the future aeronautical system for continental communication L-DACS and the Distance Measuring Equipment DME

Najett Neji, Raul de Lacerda, Alain Azoulay SUPÉLEC - DRE 91192 Gif Sur Yvette - France {surname.name}@supelec.fr

Thierry Letertre SUPÉLEC - TELECOM 91192 Gif Sur Yvette - France thierry.letertre@supelec.fr Olivier Outtier DGAC - DSNA 75015 Paris - France olivier.outtier@aviation-civile.gouv.fr

Abstract-In the last decade, the aeronautical authorities expressed their need to develop a new system for aeronautical radiocommunications. The L-band Digital Aeronautical Communication System (L-DACS) is the part of the future system that will be operating in a part of the aeronautical L-band (960-1164 MHz), already occupied by a large number of radio-frequency legacy systems. This paper aims at studying the L-DACS impact on a legacy system, the Distance Measuring Equipment (DME), under two critical situations. Such topics are fundamental in aeronautics, as any communication or radionavigation dysfunction may endanger flight and passengers security. Some obtained results will be used for the L-DACS standardization and its specifications finalization. For the first scenario (air-to-air scenario), we propose a deterministic algorithm to compute the interference level in the frequency domain. Since this seems to be insufficient for the second scenario (co-site scenario), we develop a time-frequency approach to analyze the interference using an aeronautical RFC test-bed that we implemented at Supelec.

#### I. INTRODUCTION

In the beginning of the 21th century, international aeronautical authorities expressed their need to develop a new infrastructure for aeronautical radiocommunications, because the air-traffic is increasing (see Fig. 1) and that current communication systems in the aeronautical VHF-band  $(118 - 136 \ MHz)$  between pilots and air-controllers are reaching their capacity limits. [1]

The L-band Digital Aeronautical Communication System (L-DACS) is the part of the future infrastructure that will be in charge of continental radiocommunication and it is expected to operate in a part of the aeronautical L-band  $(960 - 1164 \ MHz)$ , already occupied by a large number of radio-frequency legacy systems (see Fig. 2). Among them, one of the most important systems is the Distance Measuring Equipment (DME), which permanently evaluates the slant distance between the aircraft and ground beacons, and which uses most of the  $960 - 1164 \ MHz$  spectrum.

Consequently, it is essential to consider its radio-frequency compatibility (RFC) for the development of the future L-DACS system. Through RFC, the coexistence between L-DACS and legacy systems can be evaluated, both in ground and airborne environments. L-DACS and a legacy system are considered compatible from a RFC perspective, if L-DACS can operate correctly (with respect to its expected quality of service) in the presence of the legacy system without generating harmful interference, and vice versa. To evaluate the RFC, it is essential ti study many interference scenarios, most of them are emphasized in Fig. 3.



Fig. 1. Air traffic forecast in Europe for 2012 (from [2]).



Fig. 2. The L-Band spectral occupancy (adapted from [3])

For the moment, two candidate technologies have been preselected and they are named L-DACS1 and L-DACS2, respectively. In this paper, we particularly focus on the L-DACS1/2 effect on the performance of a DME on-board receiver. To this end, we identified two among the most critical interference scenarios from the RFC viewpoint. Under the first scenario called the air-to-air scenario, L-DACS and DME are airborne equipment onboard of distinct aircrafts whereas for the second scenario, named the co-site scenario, both airborne devices are placed in the same aircraft.

The paper is organized as follows. In Section II, we give



Fig. 3. Main interference scenarios to be studied for the RFC.

better insights on the L-DACS system and development status, and we present the DME system. In Sections III and IV, we analyze L-DACS impact on DME under the air-to-air scenario and co-site scenario, respectively. For both cases, we describe our methodology and emphasize the main results. Finally, in Section V, we summarize the paper contributions and we provide some perspectives for further research works.

#### II. L-DACS AND DME SYSTEMS PRESENTATION

Since their standardization, aeronautical communications are essentially analogical, and they are performed in the aeronautical VHF band (118 - 136 MHz). On the other hand, according to the aeronautical authorities forecasts, the aerial traffic is expected to grow continually. Consequently, a congestion phenomenon is likely to happen in the few coming years in regions of the world with the highest traffic load.

In this context, the International Civil Aviation Organization (ICAO) launched in 2004 a Euro-American initiative to develop a future infrastructure for aeronautical communications (FCI). The FCI development started under a cooperative research program named Action-Plan 17 (AP17) [4] and involves research teams, industrial partners and aeronautical authorities from many countries in the world. The first step of the FCI development activities was to determine the most promising technologies to fulfill the new aeronautical requirements expressed in the report [5]. According to the first assessment results published in 2007 [6], [7], the infrastructure comprises many technologies depending on the flight phase.

Among these technologies, the L-band Digital Aeronautical Communication System (L-DACS) was identified to be in charge of the continental communications. It is expected to operate in a part of the aeronautical L-Band (960 - 1164 MHz) [8]. The L-band is potentially large but its spectrum occupation is dense, not only by diverse aeronautical systems but also by mobile telephony and satellite navigation systems as it is shown in Fig. 2. Therefore, it is important to address the L-DACS coexistence with all these systems in both airborne and ground environments.

For the moment, two technologies have been identified as potential candidates to support L-DACS and they are named L-DACS1 [9], [10] and L-DACS2 [11], [12] . L-DACS1 is similar to the IEEE 802.16 wireless system and is based on Frequency-Duplex Division (FDD) technique, where the ground station and the airborne equipment transmit



Fig. 4. The applied process for L-DACS candidate selection (from [9] and [11].

simultaneously using distinct frequency bands. However, the L-DACS2 is similar to the GSM standard and is based on Time-Duplex Division (TDD) technique, where the ground station and the airborne equipment transmit using the same carrier frequency during distinct time intervals. We summarize in Table **??** main system parameters of both candidates.

 TABLE I

 L-DACS1 AND L-DACS2 MAIN SYSTEM PARAMETERS [9]–[12]

System Parameters	L-DACS1	L-DACS2
System range	200 NM	200 NM
Airborne cable loss	3 dBi	3 dBi
Transmitting effective bandwidth	498,05 kHz	200 kHz
Maximum ground transmit power	46 dBm	55,4 dBm
Maximum airborne transmit power	46 dBm	47 dBm
Ground cable insertion losses	2 dB	2.5 dB
Receiving effective bandwidth	498,05 kHz	200 kHz

L-DACS1/2 systems development is now a part of two research programs (SESAR [13] in Europe and NextGen [14] in the USA, additionally to parallel activities in Japan [15]) and three main tasks have been identified: systems specifications, prototypes development and performance evaluation through operational scenarios and interference scenarios. These tasks are shown in Fig. 4.

In this paper, we are interested in the third task and we aim to analyze the L-DACS1/2 impact on an L-band legacy system, the Distance Measuring Equipment (DME).

The DME [16] is a radio navigation system that is nowadays used in all airplanes to measure permanently the slant distance between an airplane and ground beacons. It operates in the 960-1215 MHz frequency band and has been used for near a century. The frequency channel used by the airplane and the one employed by the ground beacon are separated by 63 MHz. The DME communication is based on three steps. It starts on the airborne equipment (named the interrogator) which sends a stream of Gaussian-shaped pulse pairs to the ground station (called the transponder). The latter sends back the received signal to the interrogator with a certain delay, and finally, by measuring the time interval between sent and received streams, the interrogator determines its distance to the transponder.

According to its specifications, the DME transmitted signal consists of a maximum of 150 random pulse pairs per second.



Fig. 5. The structure of a DME pulse pair



Fig. 6. Horizontal and vertical aircrafts separations imposed by aeronautical instances.

In each pair, the peaks of the two gaussian pulses are separated by at least 12  $\mu s$  depending on the DME mode of operation. A DME shape is given in Fig.

#### III. AIR-TO-AIR INTERFERENCE SCENARIO ANALYSIS

Under the air-to-air scenario, L-DACS and DME systems are onboard distinct aircrafts. This scenario is critical because the victim receiver is likely to intercept signals from many interferers (this is due to its high radio visibility). Moreover, this scenario seems useful for both the aeronautical network dimensioning and frequency/distance planning.

We evaluated the L-DACS1/2 interference effect based on a frequency-domain analysis and using a specific algorithm to compute the interference level in the worst case. Under this situation, the victim receiver is likely to get the maximum interference level. In addition, we considered the case of co-channel interference, where the interferer and the victim receiver use overlapping frequency bands. Finally, we assumed a free-space propagation model.

We defined a deterministic approach, different from existing methodologies dealing with L-DACS interference, taking into account the imposed vertical and horizontal aircrafts separations by aeronautical authorities (see Fig. 6).

To model the aeronautical environment, we used L-DACS antennas characteristics (which are typical L-band antennas,



Fig. 7. Total interference spectral density at the DME victim receiver, by two L-DACS1 transmitters (for 2, 1 and 0 in-band interferers).

omnidirectional in azimuthal direction and slightly directive in vertical direction to place interferers in a bi-dimensional grid. The methodology assumes at maximum two airplanes per flight level and is based on three steps. In the first step, we identify the strongest one/two interferer(s) for each flight level independently. In the second step, we select the positions of the K identified airplanes generating the highest cumulative interference power at the victim receiver. In the last step, we allocate to each interferer a frequency channel such as the interference level remains the highest.

In previous publications [18] and [19], we applied this approach to an ideal victim receiver. In this paper, we analyze the L-DACS interference considering a DME receiver.

From their specifications, L-DACS1 uses a 0.5 MHzbandwidth for transmission/reception and its airborne transmission power is 16 dBW (46 dBm); and L-DACS2 uses a 0.2 MHz bandwidth for transmission/reception and its airborne transmission power is 17 dBW (47 dBm). The DME uses a 1 MHz bandwidth and for the DME receiver, the computed maximum acceptable interference spectral density is  $-129 \ dBW/MHz$  (-99 dBm/MHz). It should be noticed that aeronautical safety margins have to be added to obtain the operational acceptable threshold.

Based on these information, we considered K = 2 L-DACS interferers around the DME victim receiver. Three cases may occur: both interferers are in-band, or only one of them is in-band or both of them are out-of-band. We computed the cumulative interference spectral density in the three situations and the results are summarized in Fig. 7 and Fig. 8. In these figures, we present the total interference spectral density  $I_d$  at the DME receiver, with respect to vertical separation between the strongest L-DACS interferer and the DME receiver altitudes.

From these results, we can see that as expected by classical frequency sharing studies, it is necessary to have no in-band L-DACS1/2 interference in order to avoid harmful interference. We also notice that interference is principally due to the



Fig. 8. Total interference spectral density at the DME victim receiver, by two L-DACS2 transmitters (for 2, 1 and 0 in-band interferers).

first identified L-DACS interferer (which is the strongest interferer). Moreover, for both L-DACS candidates, we obtained a precise estimation of the interference level using our deterministic approach.

#### IV. CO-SITE INTERFERENCE SCENARIO ANALYSIS

For the second identified interference scenario, called the co-site scenario, both the interferer and the victim receiver are on-board the same aircraft. This scenario is considered as the most critical from a RFC viewpoint because of equipment proximity. From now on, we precisely focus on one of the L-DACS candidates on the DME performance and we chose to investigate L-DACS2.

The frequency-domain analysis seems insufficient for this case. In fact, the interference level  $P_{DME}$  received by the DME interrogator is given by equation (1):

$$P_{DME}(dBW) = P_{LDACS2}(dBW) - L_{LDACS2}(dB) - C_A(dB) - L_{DME}(dB) + M_{dB}(dB),$$
(1)

where  $P_{LDACS2}$  (in dBW) is the L-DACS2 output power,  $L_{LDACS2}$  and  $L_{DME}$  (in dB) are cable losses at the L-DACS2 transmitter and the DME interrogator, respectively, and  $C_A$  (in dB) is the coupling between their antennas.  $M_{dB}$  (in dB) is the attenuation obtained by the superposition between the L-DACS2 transmit mask and the DME blocking mask.

Now, using system characteristics given in [11], we have  $P_{LDACS2} = 17 \ dBW$  and  $L_{LDACS2} = 3 \ dB$ . In addition, according to [20],  $C_A = 20 \ dB$ . Moreover, at 1 MHz offset relatively to its center frequency, the L-DACS2 maximum mask attenuation is 80 dB (see [21]). Hence, if L-DACS2 and DME frequencies separated by 1 MHz,  $M_{dB} = -80 \ dB$ . If we assume that  $L_{DME} = 2 \ dB$ , we obtain  $P_{DME} = -88 \ dBW$ . Knowing that the DME bandpass is equal to 1 MHz, the interference density is  $I_{DME} = -88 \ dBW/MHz$  at the DME receiver interrogator input.



Fig. 9. RFC system model: DME transmitter, L-DACS2 interferer and DME victim.

However, as mentioned in [22], the maximum acceptable interference level by the DME interrogator is  $I_{max} = -129 \ dBW/MHz$  (before taking into account aeronautical margins).

Thus, to study this scenario, we consider a different methodology, named time-domain approach, considering system dynamics and technology properties. Let us first describe the interference process in this case (see Fig. 9).

The L-DACS interference is added to the DME signal sent by the transponder to the interrogator (the reply signal, which is a replica of the interrogation signal). In this situation, collision between L-DACS and DME signals may occur. Due to this phenomenon, some Gaussian-pulse pairs may be lost at the DME airborne receiver. Let us call R (in %) the rate of correctly received DME pairs, compared to the total number of sent pairs in the interrogation signal. If Rbecomes lower than a certain threshold Rmin after a precise time delay (called the maximum synchronization time), the synchronization between the DME interrogator and the DME ground beacon is considered lost.

In this paper, we implement the scenario considering an L-DACS2 interferer (TDD system), through specific test-beds with an L-DACS2 signal generator and DME commercial equipment. Our criterion of analysis is the DME synchronization state (binary information), which can be easily detected by observing the DME interrogator screen during the maximum synchronization time (see Fig. 10). The synchronization state is 1 if the distance is displayed with acceptable precision (referring to DME interrogator specifications) before the maximum synchronization time expires, and 0 else.

In the previous papers [23] and [24], we presented the obtained measurement results for the conducted mode, where L-DACS unwanted signals are propagated through cables, connectors and electronic devices. In this work, we are more focused on laboratory measurements in the radiated mode, where unwanted signals are radiated by the L-DACS2 antenna and captured by the DME antenna. For this mode, we first enumerate the used material, then we describe the experimental setup and we detail the followed protocol.



Fig. 10. Displayed information by the synchronized DME interrogator.



Fig. 11. Schematic diagram for RFC measurements in the radiated mode.

For the experimental material (see Fig. 11), we associate an antenna to the L-DACS2 transmitter, an antenna to the DME interrogator and an antenna to DME transponder. In addition to the L-DACS2 signal generator (Agilent E4438C ESG), the DME used commercial interrogator (Bendix King KN62A) and the DME commercial transponder (Aeroflex IFR6000 Ramp test), the test-bed comprises:

- Two identical <u>commercial aeronautical antennas</u> manufactured by "*Comant*" and associated to the DME interrogator (antenna 1) and the L-DACS2 generator (antenna 2),
- A specific antenna provided with the Aeroflex IFR6000 pack and associated to the DME transponder (antenna 3),
- A network/spectrum analyzer manufactured by "*Rohde* and Schwarz" (ZVL type) to measure signals at antenna ports of the different devices.

In Fig. 11, black-colored arrows indicate the power circulation starting from the DME interrogator and green-colored arrows show outputs to spectrum analyzers. We also precise the minimum and maximum output powers  $P_s$  and maximum input power  $P_e$  of our equipments as well as the minimum and maximum input power of our spectrum analyzers. We implemented this diagram with commercial equipments and antennas at the planar anechoic chamber at Suplec (see Fig. 12).

To perform the measurements, we proposed the following experimental protocol, based on the reference document [20], and composed by 11 steps <sup>1</sup>:

1- Verify the link budget before beginning measurements.





Fig. 12. Implemented aeronautical RFC test-bed in the radiated mode.

2- Switch on the DME interrogator's alimentation.

3- Turn on the DME interrogator and set its VOR frequency (based on tables from [20], this selected VOR frequency will determine a specific DME channel and consequently both interrogator and transponder center frequencies).

4- Check L-DACS2 and DME equipments' good functioning through preliminary tests.

5- Generate the L-DACS2 time-domain radio-frequency (RF) signal.

6- Activate the L-DACS2 generator "local" mode.

7- Set the L-DACS2 maximum power (depending on the target SIR value), its modulation and set its center frequency.

8- Increase the L-DACS2 channel occupation rate.

9- Run the test.

10- Count a certain time delay (corresponding to the maximum synchronization time) and observe the DME synchronization state. For each measure, we verify both stability and precision criteria. We considered that a measure is stable if the same DME synchronization state is obtained after two consecutive tests under the same conditions.

11- Stop the test.

We also assumed in-band interference and out-of-band interference. For the experimental protocol, we considered 2 seconds for the maximal synchronization time, and  $S_{min} + 3 dB$  for the useful signal power at the DME victim receiver, being  $S_{min}$  (in dB) the tested DME equipment sensitivity (both values are based on the DME specifications). Finally, we selected the DME channel 17X, which corresponds to the carrier frequency  $F_c = 978 MHz$  for the transponder (this corresponds to the lowest operational frequency for the investigated DME device). Under these assumptions, we analyze the impact of the L-DACS2 channel occupation rate, which is the percentage of time of effective L-DACS2

<sup>&</sup>lt;sup>1</sup>For a fixed power in step 7, repeat steps 8 to 11 until the DME synchronization state changes. Then, increase the power and follow the same procedure.



Fig. 13. DME receiver performance measurement in presence of L-DACS2 interference, with respect to the frequency offset between useful and interfering transmitters.

transmission, on the DME performance. We summarize the obtained results in Fig. 13, where the maximum acceptable L-DACS2 channel occupation rate is shown with respect to the frequency separation between L-DACS2 and DME transponder signal carriers, for different values of the Signal to Interference Ratio (SIR) at the DME airborne receiver.

We can see that under this scenario and with the specific tested equipment, the DME synchronization is successful in the presence of the L-DACS2 interferer if its channel occupation rate remains lower than 72 % for *SIR* higher than  $-40 \ dB$  and this also occurs in the co-channel mode (L-DACS2 and DME use overlapping channels). Hence, based on our proposed approach, implemented test bed and available DME commercial equipment at Supelec, L-DACS2 would not cause harmful interference on the DME system. We prove herein that while time-domain parameters are considered, satisfying results in terms of RFC can be obtained. In fact, according to existing RFC studies, in the co-channel mode, these systems cannot coexist.

#### V. CONCLUSION

In this paper, we studied the L-DACS interference impact on the DME receiver performance under some critical scenario from the RFC viewpoint (air-to-air scenario and co-site scenario). To analyze the first scenario, we proposed a deterministic frequency-domain methodology which is different from existing approaches mentioned in frequency sharing studies involving L-DACS, and which gives a good estimation of the interference level at the victim receiver. For the second scenario we developed a time-frequency approach, which takes into account system dynamics and which provides more realistic evaluation to the actual interference level. More particularly, for the co-site scenario, we proved that under this new methodology and using a specific laboratory setup to emulate the interference environment, L-DACS2 would not cause harmful interference on the DME interrogator, whereas in classical studies, this was not expected to happen. For further work, these results should be confirmed by additional tests with other commercial DME devices using other metrics to analyze their performance. To complete the analysis, it is also important to study the DME effect on the L-DACS2 performance using the time-frequency approach, then to extend it to other interference scenarios, in order to evaluate the interference more precisely than under classical RFC methodologies. We finally suggest generalizing the proposed time-frequency approach to analyze the RFC between any two radio-frequency systems, taking into account additional parameters related to system dynamics and their technology properties.

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