

Fast Calculation of Response of Scatterers in Uniaxial Laminates

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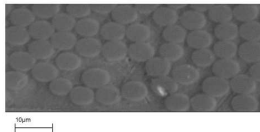
Outline

- 1 Introduction
- 2 Fast and stable calculation of spectral responses
- 3 Fast calculation of impedance matrix
- 4 Conclusion

Introduction

- Layered structures are of great interest in non-destructive testing (NDT) community due to their vast practical implementations in many areas.
- Nowadays, more and more composite materials have been used to construct such layered structures due to their lightness and robustness, such as the carbon fiber reinforced materials.
- However, due to the intrinsic anisotropy of some composite materials, the NDT problems with these materials become challenging.

A schematic of fibres in composite materials from [A. Galehdar, *et. al.*, Proc. the 40th European Microwave Conference, pp. 882-885, 2010, Paris, France.]



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Introduction

- A proper and efficient modeling of the layered anisotropic media is essential for tackling detection problems. The **volume integral equation method** is a good candidate.
- The **full vectorization** of the propagator matrix method will be introduced to give the spectral response of anisotropic laminates due to **any bounded active source** within the laminates.
- In order to achieve good efficiency and accuracy, a new **windowing technique** is introduced. Furthermore, interpolation and integration algorithms based on the **Padua points** are implemented.

State Equation

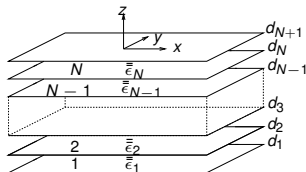
- Uniaxial permittivity tensors:

$$\bar{\epsilon}'_n = \text{diag} \left[\epsilon_{11}^{(n)}, \epsilon_{22}^{(n)}, \epsilon_{22}^{(n)} \right]$$

with the optical axes always rotating in the $x - y$ plane.

- In the Fourier domain, the field vector is defined as

$$\bar{\varphi}(k_x, k_y, z) = \begin{bmatrix} k_x \tilde{H}_x(k_x, k_y, z) + k_y \tilde{H}_y(k_x, k_y, z) \\ k_y \tilde{H}_x(k_x, k_y, z) - k_x \tilde{H}_y(k_x, k_y, z) \\ k_x \tilde{E}_x(k_x, k_y, z) + k_y \tilde{E}_y(k_x, k_y, z) \\ k_y \tilde{E}_x(k_x, k_y, z) - k_x \tilde{E}_y(k_x, k_y, z) \end{bmatrix}$$



- The state equation reads (for each (k_x, k_y))

$$\frac{d}{dz} \bar{\varphi}(z) = \bar{A}_n \cdot \bar{\varphi}(z) + \bar{f}(z)$$

with $\bar{A}_n = \bar{U} \cdot \bar{\Sigma} \cdot \bar{U}^{-1}$ for the n^{th} layer (can be constructed analytically) and $\bar{f}(z)$ the source term.

Solution of the state equation

- The solution reads

$$\bar{\varphi}(d_{n+1}) = e^{\bar{\bar{A}}_n(\delta_n)} \cdot \bar{\varphi}(d_n) + \int_{d_n}^{d_{n+1}} e^{\bar{\bar{A}}_n \delta'_n} \cdot \bar{f}(z') dz'$$

- Cannot be directly calculated due to the numerical instability, since

$$e^{\bar{\bar{A}}_n(\delta_n)} = \bar{\bar{U}} \cdot e^{\bar{\bar{\Sigma}}(\delta_n)} \cdot \bar{\bar{U}}^{-1}, \text{ and } e^{\bar{\bar{\Sigma}}(\delta_n)} \text{ could explode when}$$

- 1 Large thickness δ_n .
 - 2 Large lateral spatial frequency k_x or k_y (representing a fast changing evanescent wave).
- For such numerical instability, wave mode decomposition method has been proposed to deal with source free problems [1] by expanding the field vector as

$$\bar{\varphi}(d_n) = \bar{\bar{\Omega}}_n \cdot [\alpha_n, \beta_n]^T, N + 1 > n > 0$$

1. H. D. Yang, IEEE Trans. Antennas Propagat., vol. 45, pp. 520-526, 1997.

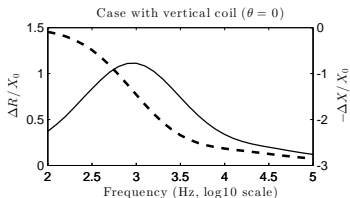
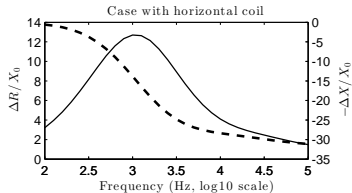
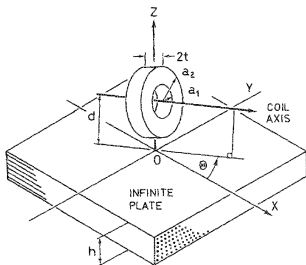


Stable solution for problems with sources

- If there is an active source embedded inside the layered media (or in the outer half space), the field transformation is different from the case without source.
- One needs to write the field vector after the transformation as $\bar{\varphi}(d_{n+1}) = \bar{\bar{\Omega}}_{n+1} \cdot [\alpha_{n+1}, \beta_{n+1}]^T + \bar{h}$.
The difference is that the constant term \bar{h} is added.
- Keys to generate an accurate source term \bar{h} :
 - To have the Fourier spectrum of the distribution of the current density.
 - To follow a stable transforming scheme similar to the one for the source-free case to avoid the stability issue.
- This constructs the spectral responses of the laminates to the current basis used in MoM.

A test

A numerical example from [1] is reproduced:



1. S. Burke, J. Appl. Phys., vol. 68, pp. 3080-3090, 1990.



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Spatial responses of uniaxial laminates

- The most time-consuming part of using MoM is to construct the **impedance matrix** from the spectral responses.
- The convention is to use the **Sommerfeld integral** to calculate the responses of the current basis at each testing point (suppose the delta function is used as testing function).
- When dealing with volume integral equation, one usually has **periodic discretization**, i.e., the **rectilinear meshing**.
- Using such meshing, one may be able to **directly construct the discrete spectrum of the impedance matrix without the Sommerfeld integral**.

Spatial responses of uniaxial laminates

- Using IFT, the response of a current basis at some test point can be obtained as

$$\eta_{u,v;p;p'}(m, n) = \text{IFT}\{\tilde{\eta}_{u,v;p;p'}(k_x, k_y)\}|_{x=x_m, y=y_n}$$

where the $\tilde{\eta}_{u,v;p;p'}(k_x, k_y)$ is the spectral response of the layers due to the current basis.

- Using the periodic property of the testing point, one can construct it in another way, by generalized Poisson summation formula:

$$\hat{\eta}_{u,v;p;p'}(\alpha, \beta) = \iint_{-\infty}^{+\infty} \tilde{\eta}_{u,v;p;p'}(k_x, k_y) \tilde{Q}_{\alpha,\beta}(k_x, k_y) dk_x dk_y$$

where $\tilde{Q}_{\alpha,\beta}(k_x, k_y)$ is the superposition of the spectrums of some window function.

Spatial responses of uniaxial laminates

- The $\tilde{Q}_{\alpha,\beta}(k_x, k_y)$ is found to be

$$\tilde{Q}_{\alpha,\beta}(k_x, k_y) \propto \sum_{l_1=-\infty}^{+\infty} \sum_{l_2=-\infty}^{+\infty} \tilde{W} \left[\frac{2\pi}{\Delta x} \left(\frac{\alpha}{2M-1} + l_1 \right) - k_x, \frac{2\pi}{\Delta y} \left(\frac{\beta}{2N-1} + l_2 \right) - k_y \right]$$

where $\hat{W}(k_x, k_y)$ is the spectrum of the chosen window function.

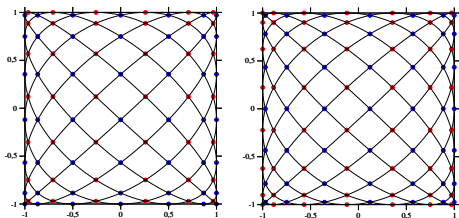
- By only integrating areas where the main lobes of the window spectrum cover, the efficiency of such an integral can be significantly increased.
- The computational efficiency ratio between the proposed method and IFT is (if the same numerical integration scheme adopted)

$$\gamma = \frac{16}{(2M-1)(2N-1)}$$

where M and N are the total meshing cells along x and y directions.

Numerical integration

- The Padua point based numerical integration method will be used.
- Make an unisolvent point set (unique interpolating polynomial) with minimal growth of their Lebesgue constant [1].
- specifically designed partition of integration area in spectral domain to fully reuse all spectral samplings.
- Computational cost $16 \times \sum_{j=1}^{M_t N_t} O(N_j^2 \log N_j)$ for all meshing points in one plane, with N_j the order of Chebyshev Polynomial used in the j^{th} partition.



$(-\cos[(n+1)t], -\cos(nt)), t \in [0, \pi], \text{ for } n = 12 \text{ and } n = 13.$

1. L. Bos, M. Caliari, M. Vianello, S. De Marchi, and Y. Xu, J. Approx. Theory, vol. 143, pp. 15-25, 2006.

Numerical test 1

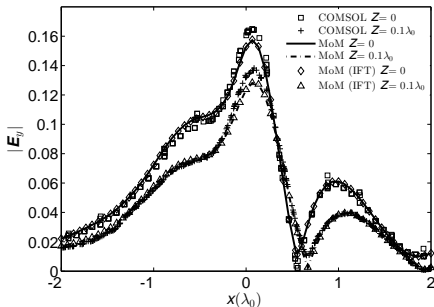
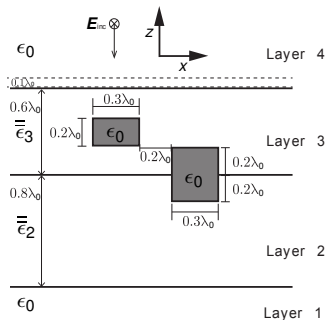
Uniaxial layered media with two air cubes inside:

$\bar{\epsilon}_2 = \text{diag} [3, 2, 2] \epsilon_0$ with 30 degree rotation angle.

$\bar{\epsilon}_3 = \text{diag} [4, 2.5, 2.5] \epsilon_0$, with 60 degree rotation angle.

thickness of both scatterers along the y direction is $0.3 \lambda_0$.

plane wave normal incidence with y polarization.



Numerical test 2

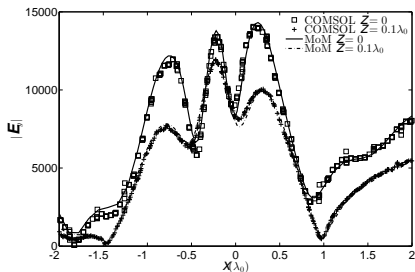
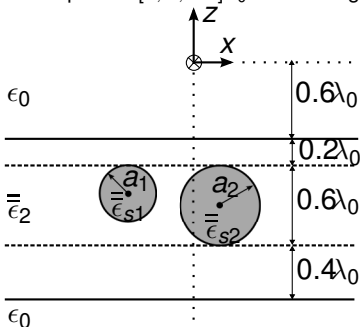
Uniaxial layered media with two spheres inside:

$\bar{\epsilon}_2 = \text{diag} [5, 2, 2] \epsilon_0$ with 60 degree rotation angle.

$\bar{\epsilon}_{s1} = \text{diag} [1, 1, 1] \epsilon_0$ and $\bar{\epsilon}_{s2} = \text{diag} [3, 1, 1] \epsilon_0$ with 30 degree rotation angle.

two spheres are of radii $0.2\lambda_0$ and $0.3\lambda_0$.

y-oriented dipole at $[0, 0, 0.6]\lambda_0$ illuminating the scatterers.



Conclusion

- A fast solution to solve scattering problems from inhomogeneities embedded in uniaxial layered media is presented, which includes
 - 1 An efficient and stable method to construct the spectral responses of the uniaxial laminates with optical axes in $x - y$ plane;
 - 2 A fast and accurate method to calculate the spatial responses of the laminates on a rectilinear mesh.
- Next step is for NdT problems involving uniaxial materials: either for inspection problems or for imaging problems.
- Further information can be found in our recently papers:
 - 1 Y. Zhong, M. Lambert, D. Lesselier, and X. Chen, IEEE Trans. Antennas Propagat., Vol. 62, pp. 247-256, 2014.
 - 2 Y. Zhong, M. Lambert, P. Ding, D. Lesselier, and X. Chen, submitted to IEEE Trans. Antennas Propagat, 2014.

Thank you!

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